

# Engineering Materials

## Problem #1:

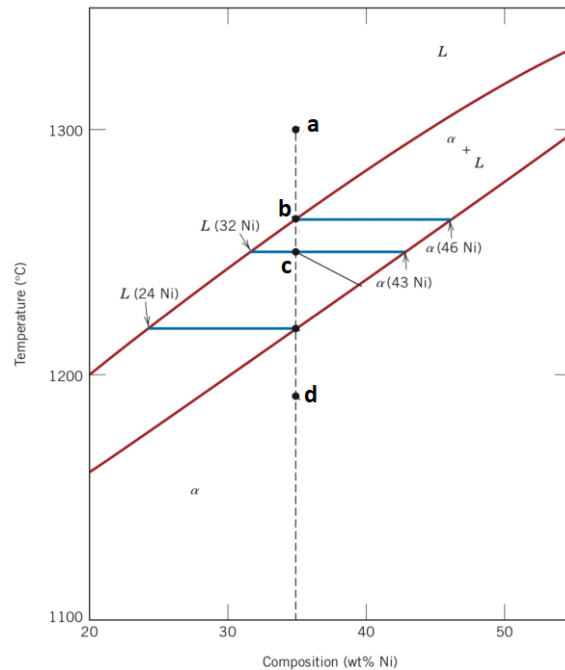
The net potential energy  $U$  between two adjacent ions is sometimes approximated by the expression

$$U(r) = -\frac{C}{r} + D e^{-r/\rho},$$

In which  $r$  is the interionic separation and  $C$ ,  $D$ , and  $\rho$  are constants whose values depend on the specific material. If  $U_0$  and  $r_0$  are the bonding energy and equilibrium separation. Derive an expression for  $U_0$  in terms of  $r_0$ ,  $D$ , and  $\rho$ .

## Problem #2:

- For a binary alloy of composition  $C_0$  in a two-phase region (i.e., phases  $\alpha$  and  $\beta$ ), the lever rule describes the mass fraction of each of the phases in terms of the phase compositions. Letting  $C_\alpha$  and  $C_\beta$  be the phase compositions and  $W_\alpha$  and  $W_\beta$  be the corresponding phase fractions, derive the expression for the lever rule [Hint: think about conservation of mass].
- A section of the Cu-Ni phase diagram is shown in the Figure to the right. Starting with an alloy with Ni composition of 35wt.% that is cooled from the liquid state at point **a** through the sequence **a-b-c-d** [see Figure to the right], draw schematics of representative microstructures at points **b**, **c**, and **d**. Assume equilibrium cooling.



- Using the Figure to the right, for the tie line that passes through point **c**, what is the mass fractions of the liquid (L) and solid ( $\alpha$ ) phases?

### **Problem #3:**

- Define the fracture toughness.
- An aircraft component is fabricated from an aluminum alloy that has a plane strain fracture toughness of  $35 \text{ MPa} \sqrt{m}$ . It has been determined that fracture results at a stress of 250 MPa when the maximum (or critical) internal crack length is 2.0 mm. For this same component, alloy, and loading configuration, will fracture occur at a stress level of 325 MPa when the maximum internal crack length is 1.0 mm? Show your work. Some relevant equations:  $K_{IC} = Y\sigma\sqrt{\pi a}$  ;  $\sigma_c = \left(\frac{2E\gamma_s}{\pi a}\right)^{1/2}$  ; and  $\dot{\epsilon}_s = K\sigma^n$ .

### **Problem #4:**

- Describe Fick's first and second laws.
- For a steel alloy, it has been determined that a carburizing heat treatment (10 h-duration) at a temperature  $T=600 \text{ K}$  will raise the carbon concentration to 0.4 wt% at a distance 1 mm from the surface. Estimate the time (in h) necessary to achieve the same concentration at a distance 3 mm for an identical steel at a temperature  $T=700 \text{ K}$ . The activation energy is  $Q=0.8 \text{ eV}$ . Reminder: The solution of the diffusion equation for these boundary conditions is given by:

$$\frac{C_x - C_0}{C_s - C_0} = 1 - \text{erf}\left[\frac{x}{2\sqrt{Dt}}\right]$$
$$D = D_0 \exp\left(-\frac{Q}{kT}\right)$$

### **Problem #5:**

- Explain the difference between thermoplastics, thermoset, and elastomers, and provide an example polymer for each.
- When plastics undergo long-term loading, they exhibit two important properties, creep and stress relaxation. Explain these phenomena and propose a test to measure each.
- What is ductile-to-brittle transition in metallic materials? What is its cause? What metals/alloys are prone to this transition?

### **Problem #6:**

Draw an A-B binary phase diagram containing one eutectic transformation and two (limited-solubility) terminal solutions,  $\alpha$  and  $\beta$ . Make sure the drawing is large enough so that it could be fully labeled.

- Label the axes and the melting points for A and B.
- Label all phase fields and identify the phases.
- Label the eutectic temperature and the eutectic composition.
- Label the point at which the solubility of B in A is maximal.

### **Problem #7:**

The behavior of a material is described by

$$\sigma = K(\varepsilon + 0.02)^n \text{ [MPa]}$$

and

$\varepsilon_f$  is the fracture strain.

- Determine Young's modulus for this material? [Hint:  $E = \lim_{\varepsilon \rightarrow 0} \left( \frac{d\sigma}{d\varepsilon} \right)$ ]
- Derive an expression for the material toughness.

### **Problem #8:**

- Clearly define the concepts of a "slip plane" and a "slip direction".
- Briefly describe the main crystallographic characteristics of the slip planes and the slip directions.
- Draw an FCC and a BCC unit cell; identify one of the slip planes in each structure, and one of the slip directions in each of the chosen slip planes.
- Explain how the number of slip planes in a material affects its ductility.

### **Problem #9:**

Unidirectional and continuous glass fibers reinforce a nylon matrix.

- Draw the stress-strain diagram for the composite; assume loading is parallel to the fiber direction (defined as  $0^\circ$  orientation), and strain extends beyond the point where the matrix deforms, causing the reinforcement to carry the entire load. Label the matrix deformation point.
- Assuming the load is applied in this direction and fibers are rigidly bonded to the matrix (no relative slip), the composite strain  $\varepsilon_c$ , the fiber strain  $\varepsilon_f$ , and the matrix strain  $\varepsilon_m$  can be considered equal. Derive an expression for the composite modulus of elasticity  $E_{c,0^\circ}$  as a function of the individual fiber and matrix moduli  $E_f$  and  $E_m$  and their respective volume fractions  $f_f$  and  $f_m$ , where  $f_f + f_m = 1$ . Begin by expressing the force in the composite as a sum of loads carried by the fiber and matrix:  $F_c = F_f + F_m$
- Now consider the case where the composite is loaded perpendicular to the fiber direction (defined as  $90^\circ$  orientation). Here it can be assumed that the stresses in the composite, fiber and matrix are equal ( $\sigma_c = \sigma_f = \sigma_m$ ). Derive an expression for the composite modulus of elasticity  $E_{c,90^\circ}$  as a function of the individual fiber and matrix moduli  $E_f$  and  $E_m$  and their respective volume fractions  $f_f$  and  $f_m$ , where  $f_f + f_m = 1$ .